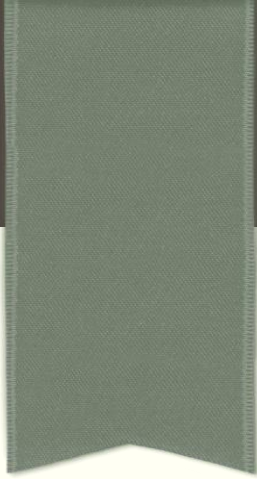




CE 261: Fluid Mechanics

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CHAPTER 3

KINEMATICS OF FLUID FLOW

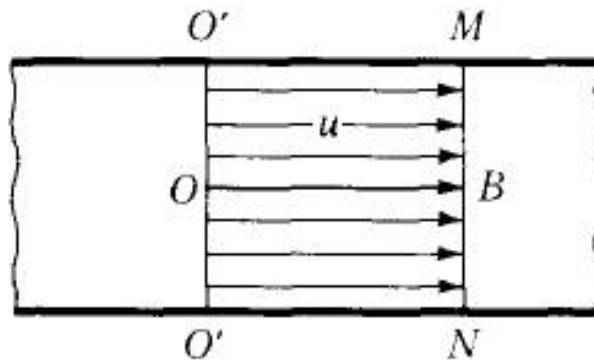
TYPES OF FLUID

- Ideal fluid
- Real fluid
- Compressible fluid
- Incompressible fluid
- Newtonian fluid
- Non-Newtonian fluid

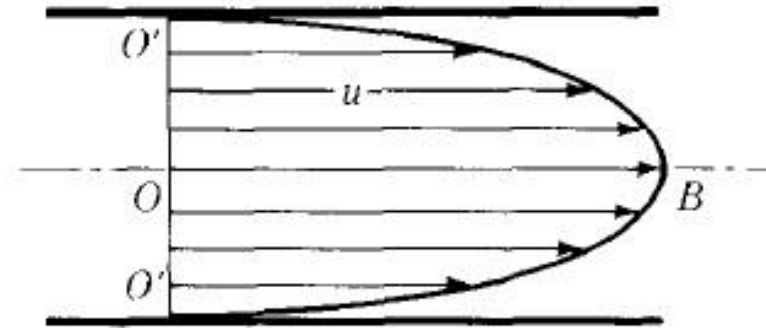
TYPES OF FLUID

Ideal Fluid: has no **Viscosity**

Real Fluid: Whenever **motion** takes places, **shearing forces** are developed



(a) Ideal fluid.



(b) Real fluid.

Figure 3.1. Typical velocity profiles. (a) Ideal fluid. (b) Real fluid.

TYPES OF FLUID

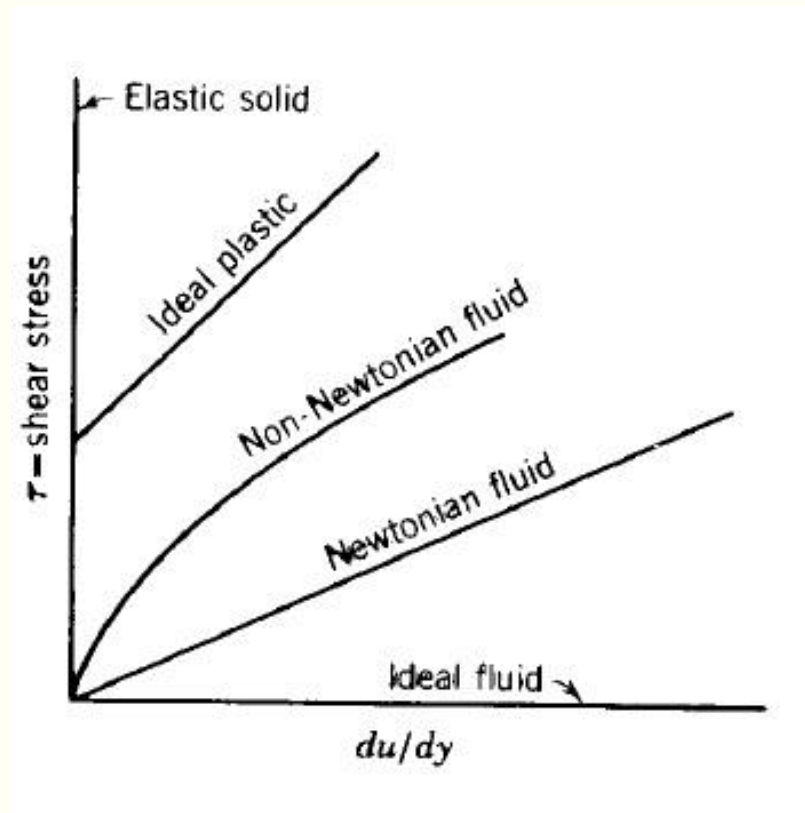
Incompressible fluid: Fluid with **constant density** with **change in pressure**

Compressible fluid: Fluid with **variable density**

TYPES OF FLUID

Newtonian fluid: A fluid for which **viscosity does not change** with **rate of deformation**

Non-Newtonian fluid: Under force it becomes more liquid or more solid.



TYPES OF FLOW

- Laminar flow and Turbulent flow
- Steady and Unsteady Flow
- Uniform and Non-Uniform Flow
- One, Two and Three Dimensional Flow

REYNOLD'S NUMBER

Reynold's experiment: <https://youtu.be/pae5WrmDzUU>

$$Re = \frac{\textit{Inertia Force}}{\textit{Viscous Force}}$$

$$Re = \frac{\rho V L}{\mu}$$

TYPES OF FLOW

Laminar flow: Type of fluid flow in which the fluid travels smoothly or in regular paths.

Turbulent flow: fluid undergoes irregular fluctuations and mixing

TYPES OF FLOW

Steady Flow: Flow properties remain **constant with respect to time**

Unsteady Flow: Flow properties **vary with respect to time.**

TYPES OF FLOW

Uniform Flow: If the flow velocity is assumed to have the same speed and direction at every point within the fluid, it is said to be uniform.

Non-Uniform Flow: If at a given instant, the velocity is not the same at every point, the flow is non-uniform.

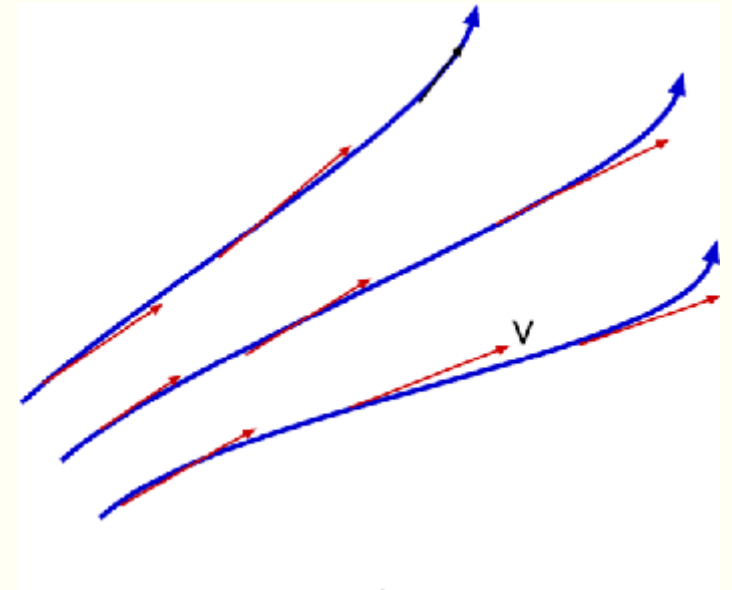
FLOW PATTERN

- Streamline
- Streamtube
- Pathline
- Streakline

FLOW PATTERN

Streamline

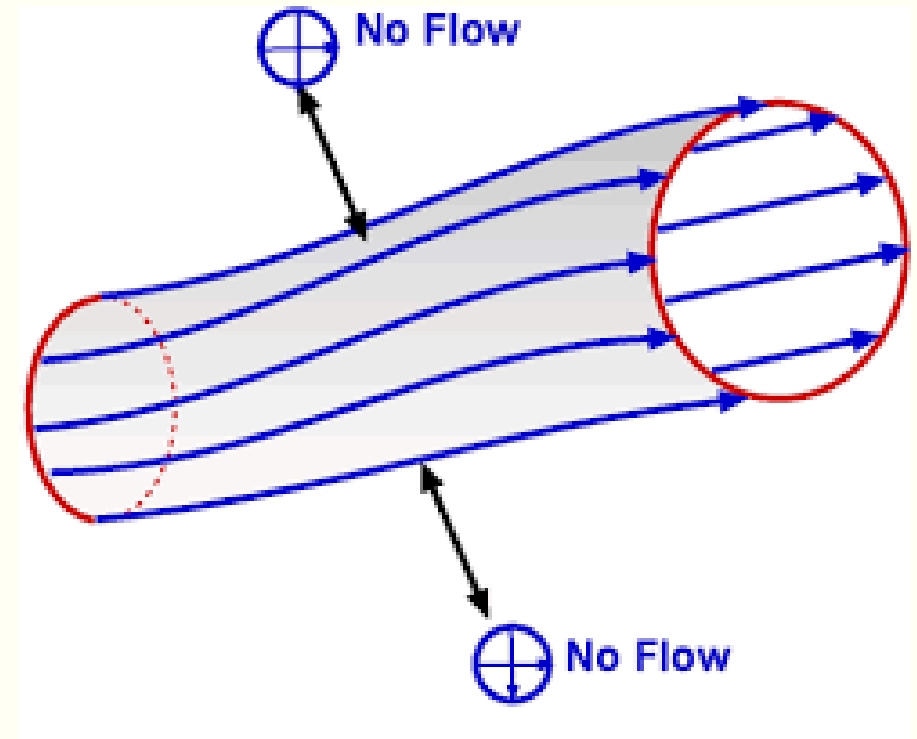
- A line which is everywhere **tangent to the velocity vector** at a given instant.
- It shows the mean direction of a number of particles at the same instant of time.



FLOW PATTERN

Streamtube:

- A bundle of streamline is called streamtube
- A streamtube is formed by a close collection of streamlines.
- Fluid can not flow in a direction perpendicular to the streamline
- Streamtube surface need not to be solid but may be fluid surface



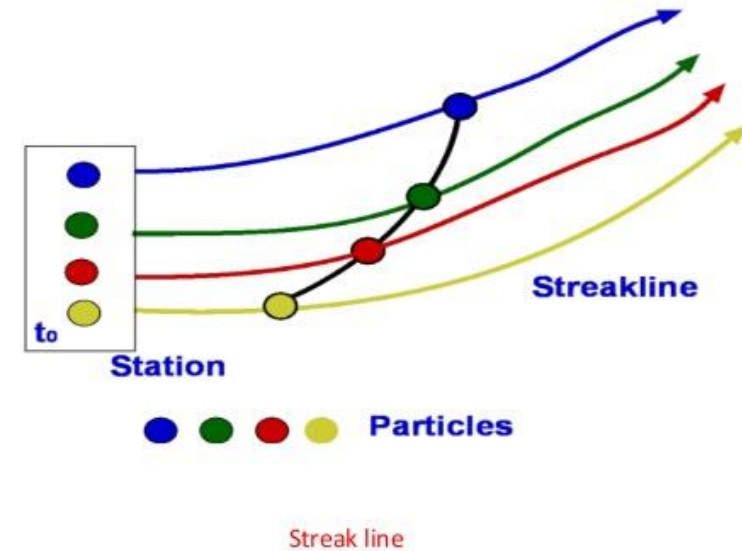
FLOW PATTERN

Pathline

- Is the trace made by a single particle over a period of time

Streakline:

Is the locus of a particle which earlier passed through a fixed point.



Continuity Equation

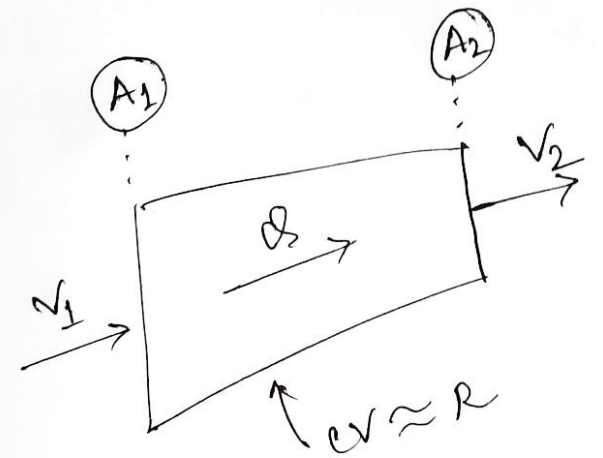
Expresses the conservation of Mass

M_t = mass of fluid contained in the control volume at time t

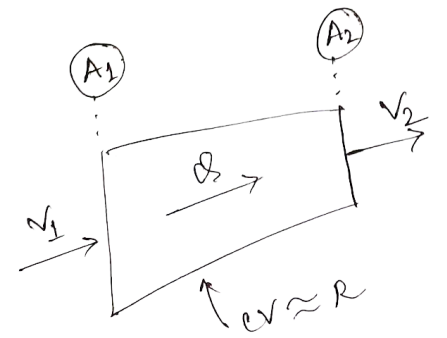
M_{t+dt} = mass of fluid contained in the control volume at time $t+dt$

$$M_{t+dt} = M_t + (\rho_1 v_1 dA_1) dt - (\rho_2 v_2 dA_2) dt$$

$$M_{t+dt} = M_t + \left(\frac{\delta \rho}{\delta t} \right) dt (R)$$



Continuity Equation



$$M_1 + (\rho_1 v_1 dA_1) dt - (\rho_2 v_2 dA_2) dt = M_2 + \left(\frac{\partial \rho}{\partial t} \right) dt (R)$$

$$\rho_1 v_1 dA_1 - \rho_2 v_2 dA_2 = \frac{\partial \rho}{\partial t} \cdot (R)$$

$$\rho_1 \int_{A_1} v_1 dA_1 - \rho_2 \int_{A_2} v_2 dA_2 = \int_R \frac{\partial \rho}{\partial t} dR.$$

Problems on Continuity Equation

A pipeline carries oil (relative density 0.86) at $v = 2$ m/s through 200mm cross-section. Find

- Mass flow rate
- Weight flow rate
- Volume flow rate
- Velocity in 60mm cross-section.

Problems on Continuity Equation

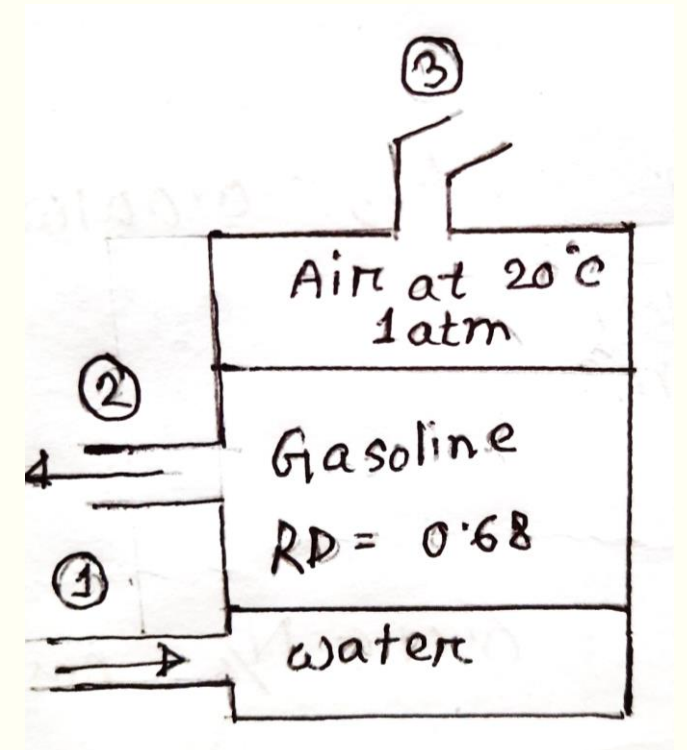
In steady viscous flow through a circular pipe the velocity profile is approximated by,

$$u = Um[1 - (\frac{r}{r_0})^2],$$

Compute the average velocity if the density is constant.

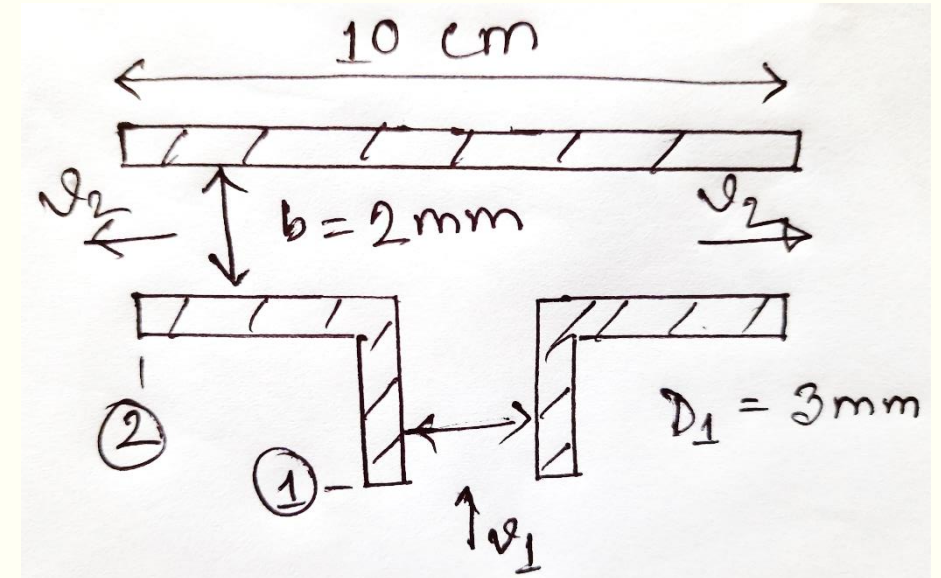
Problems on Continuity Equation

The tank in the figure admitting water at 90 N/s and ejecting gasoline (RD = 0.68) at 50 N/s. If all the fluids are incompressible, how much air is passing through the vent and in what direction? $\gamma_{\text{air}} = 0.0118 \text{ KN/m}^3$



Problems on Continuity Equation

Oil enters at section 1 at 0.06 N/s to lubricate the thrust bearing. The 10 cm diameter bearing plates are 2 mm apart. Assume steady flow compute the inlet velocity V_1 and outlet velocity V_2 .



Velocity and acceleration in steady flow

$$u_{st} = u(x, y, z)$$

$$v_{st} = v(x, y, z)$$

$$w_{st} = w(x, y, z)$$

$$\mathbf{a}_{st} = \frac{d}{dt} \mathbf{V}(x, y, z) = \frac{\partial \mathbf{V}}{\partial x} \frac{dx}{dt} + \frac{\partial \mathbf{V}}{\partial y} \frac{dy}{dt} + \frac{\partial \mathbf{V}}{\partial z} \frac{dz}{dt}$$

$$|\mathbf{V}| = (u^2 + v^2 + w^2)^{1/2}$$

$$\mathbf{a}_{st} = u \frac{\partial \mathbf{V}}{\partial x} + v \frac{\partial \mathbf{V}}{\partial y} + w \frac{\partial \mathbf{V}}{\partial z}$$

$$(a_x)_{st} = u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} + w \frac{\partial u}{\partial z}$$

$$(a_y)_{st} = u \frac{\partial v}{\partial x} + v \frac{\partial v}{\partial y} + w \frac{\partial v}{\partial z}$$

$$(a_z)_{st} = u \frac{\partial w}{\partial x} + v \frac{\partial w}{\partial y} + w \frac{\partial w}{\partial z}$$

Velocity and acceleration in unsteady flow

$$a_x = \left(u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} + w \frac{\partial u}{\partial z} \right) + \frac{\partial u}{\partial t}$$

$$a_y = \left(u \frac{\partial v}{\partial x} + v \frac{\partial v}{\partial y} + w \frac{\partial v}{\partial z} \right) + \frac{\partial v}{\partial t}$$

$$a_z = \left(u \frac{\partial w}{\partial x} + v \frac{\partial w}{\partial y} + w \frac{\partial w}{\partial z} \right) + \frac{\partial w}{\partial t}$$

Problems

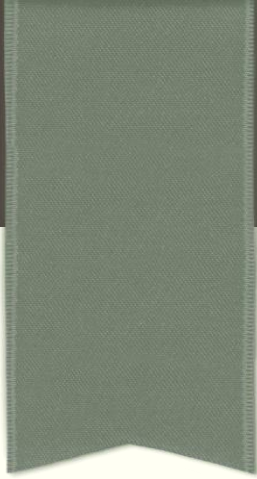
A two dimensional flow field is given by $u = 2y$, $v = x$. Derive a general expression for the velocity and acceleration. Find the acceleration in the flow field at point A ($x = 3.5$, $y = 1.2$)

$$V = (u^2 + v^2)^{1/2} = (4y^2 + x^2)^{1/2}$$

$$a_x = u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} = 2y(0) + x(2) = 2x$$

$$a_y = u \frac{\partial v}{\partial x} + v \frac{\partial v}{\partial y} = 2y(1) + x(0) = 2y$$

$$a = (a_x^2 + a_y^2)^{1/2} = (4x^2 + 4y^2)^{1/2}$$



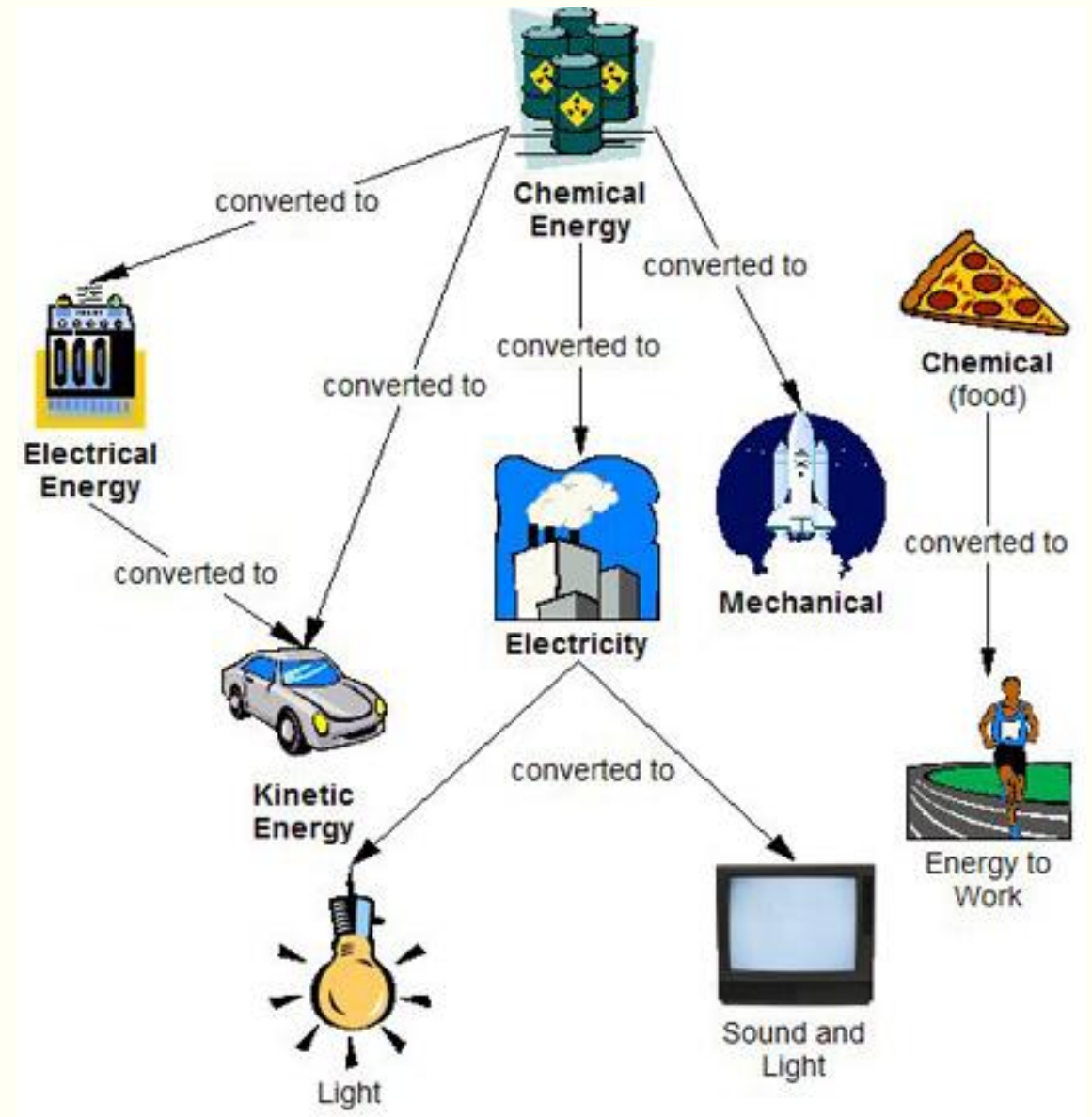
CHAPTER 4

ENERGY CONSIDERATION IN STEADY FLOW

First law of thermodynamics
says —

**Energy can neither be
created nor destroyed**

All forms of energy are
equivalent



Various Forms of energy

Kinetic Energy of Flowing Fluid

A body of mass m flowing with a velocity V possesses a kinetic energy,

$$KE = \frac{1}{2}mv^2$$

$$\frac{KE}{Weight} = \frac{\frac{1}{2}mv^2}{mg}$$

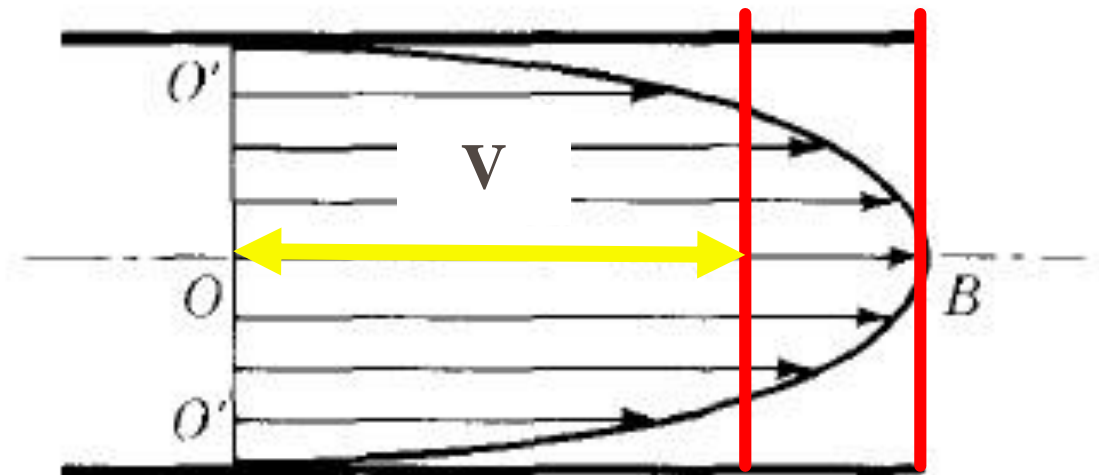
$$\frac{KE}{Weight} = \frac{v^2}{2g}$$

Various Forms of energy

Kinetic Energy of Flowing Fluid

The velocities of the different **real fluid particles** are not the same, so it is necessary to integrate **all velocity component along any direction** of the stream to obtain **the true value of the kinetic energy**

$$\frac{KE}{Weight} = \alpha \frac{V^2}{2g}$$



(b) Real fluid.

Various Forms of energy

Kinetic Energy of Flowing Fluid

To obtain the expression for α , let's consider velocity component u vary along the vertical direction.

The mass flow per unit of time through an elementary area dA is $\rho dQ = \rho u dA$

Flow of kinetic energy per unit of time across area $dA = \frac{1}{2} \rho u dA u^2 = \frac{1}{2} \rho u^3 dA$

Weight flow rate through $dA = \gamma Q = \gamma u dA$

$$\frac{KE/time}{Weight/time} = \frac{\frac{1}{2g} \gamma u^3 dA}{\gamma u dA} = \frac{\frac{1}{2g} u^3 dA}{u dA} = \frac{\frac{1}{2g} \int u^3 dA}{\int u dA}$$

$\alpha = 2$, for laminar flow in circular pipe

$\alpha = 1.01 - 1.15$, for turbulent flow

$$\alpha \frac{V^2}{2g} = \frac{\frac{1}{2g} \int u^3 dA}{\int u dA} \quad \alpha = \frac{\int u^3 dA}{V^2 \int u dA} = \frac{1}{AV^3} \int u^3 dA$$

Average of cube is greater than the cube of the average, the value of α will always be greater than 1

Various Forms of energy

Potential Energy of Flowing Fluid

Depends on elevation above any arbitrary datum.

A fluid particle of weight W situated a distance z above datum has potential energy of

$$PE = Wz$$

$$\frac{PE}{Weight} = \frac{Wz}{W} = z$$

Various Forms of energy

Internal Energy of Flowing Fluid

It is a function of **temperature**.

Since we are concerned only with difference of energy, so $\Delta i = C_v \Delta T$

Where C_v = specific heat at constant volume.

specific heat at constant volume, C_v : the amount of heat that is required to raise the temperature of unit mass of gas by 1 degree at **constant volume**.

General Equation for Steady Flow of Any Fluid

First Law of thermodynamics states:

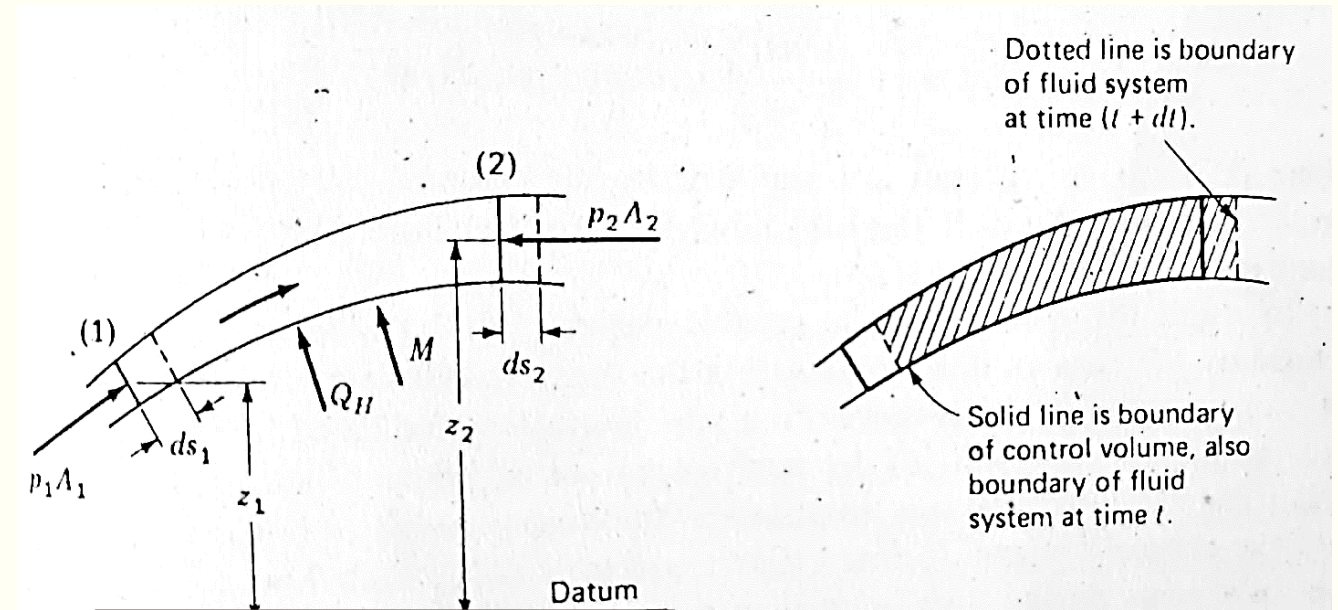
For *steady flow* the external work done on any system plus the thermal energy transferred into or out of that system is equal to the change of energy of the system

$$\text{Work} + \text{Heat} = \Delta \text{Energy}$$

$$\gamma_1 A_1 ds_1 = \gamma_2 A_2 ds_2$$

$$\text{Flow work} = p_1 A_1 ds_1 - p_2 A_2 ds_2$$

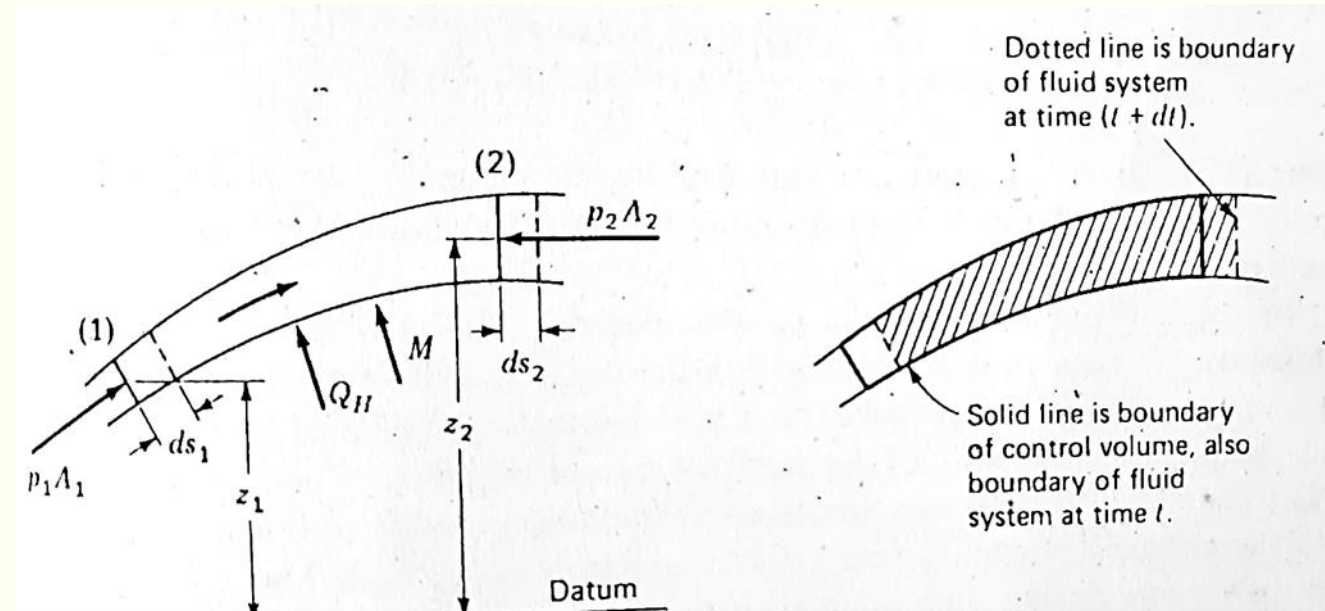
h_m is the energy put into the flow by machine per unit weight of flowing fluid



General Equation for Steady Flow of Any Fluid

$$\text{Heat} = \left(\gamma_1 A_1 \frac{ds_1}{dt} \right) Q_H dt = (\gamma_1 A_1 ds_1) Q_H$$

$$\gamma_1 A_1 ds_1 \left(z_1 + \alpha V_1^2 / 2g + I_1 \right),$$



$$\Delta \text{ energy} = \Delta E = \gamma_2 A_2 ds_2 \left(z_2 + \alpha_2 \frac{V_2^2}{2g} + I_2 \right) - \gamma_1 A_1 ds_1 \left(z_1 + \alpha_1 \frac{V_1^2}{2g} + I_1 \right)$$

General Equation for Steady Flow of Any Fluid

$$\frac{p_1}{\gamma_1} - \frac{p_2}{\gamma_2} + h_M + Q_H = \left(z_2 + \alpha_2 \frac{V_2^2}{2g} + I_2 \right) - \left(z_1 + \alpha_1 \frac{V_1^2}{2g} + I_1 \right)$$

$$\left(z_1 + \frac{p_1}{\gamma_1} + \alpha_1 \frac{V_1^2}{2g} + I_1 \right) + h_M + Q_H = \left(z_2 + \frac{p_2}{\gamma_2} + \alpha_2 \frac{V_2^2}{2g} + I_2 \right)$$

General Energy Equation

Bernoulli's Equation

$$\left(z_1 + \frac{p_1}{\gamma_1} + \alpha_1 \frac{V_1^2}{2g} + I_1 \right) + h_M + Q_H = \left(z_2 + \frac{p_2}{\gamma_2} + \alpha_2 \frac{V_2^2}{2g} + I_2 \right)$$

$$Z + \frac{P}{\gamma} + \frac{V^2}{2g} = \text{constant}$$

Assumptions:

- **Non viscous** fluid
- **Incompressible** fluid
- Uniform Velocity Distribution
- No mechanical energy added along the streamline

Head

Each term in the Bernoulli's equation has the dimension of length.

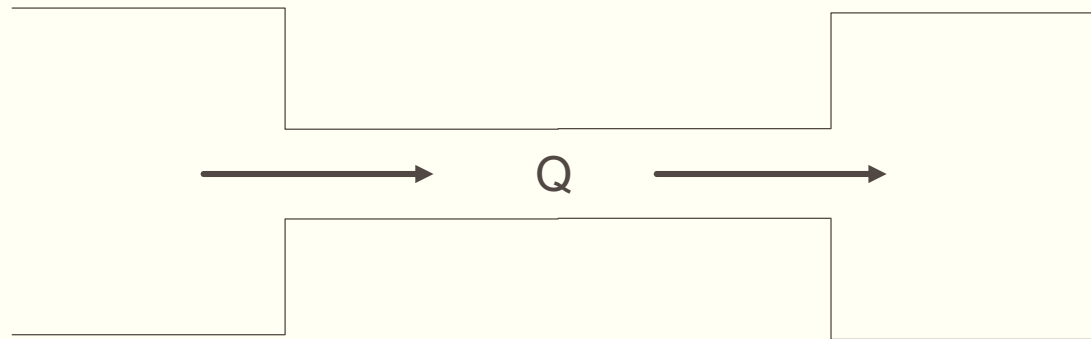
➤ Pressure Head = $\frac{P}{\gamma}$

➤ Velocity Head = $\frac{V^2}{2g}$

➤ Elevation Head = Z

$$\text{Total head, } H = Z + \frac{P}{\gamma} + \frac{V^2}{2g}$$

Problems on Bernoulli's Equation



Problems on Bernoulli's Equation

Consider Flow through a horizontal conical diffuser with following details:

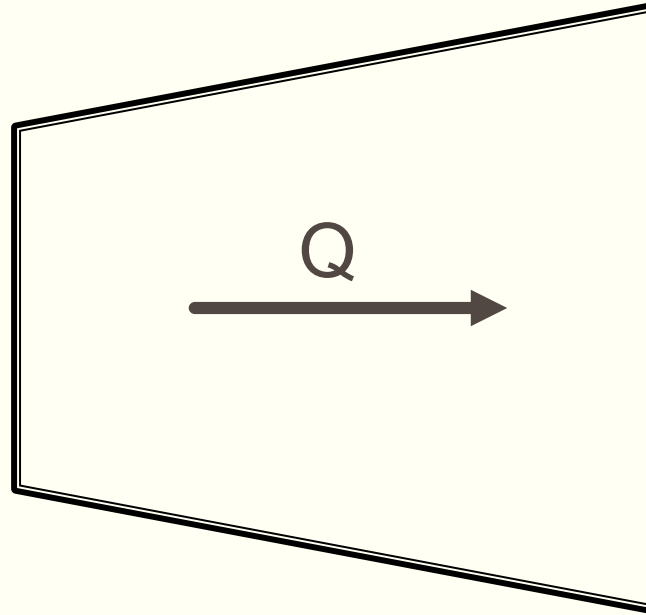
$$Q = 5 \text{ m}^3/\text{s}$$

$$D1 = 1.2\text{m}$$

$$D2 = 1.6\text{m}$$

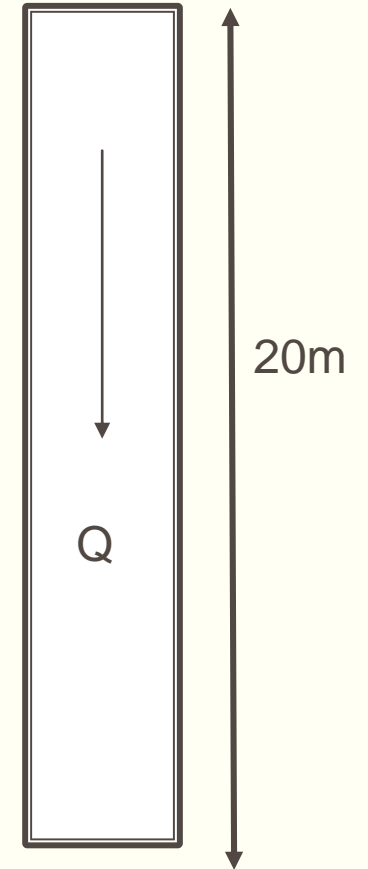
$$P1 = 7.5 \text{ kN/m}^2$$

$$P2 = ?$$



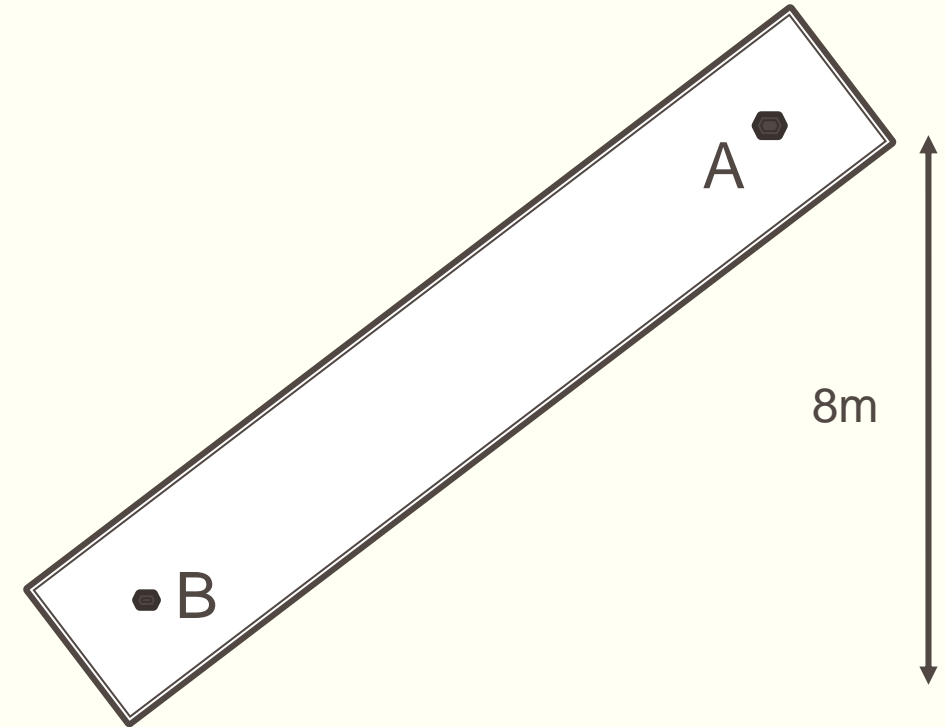
Problems on Energy Equation

Upstream pressure head and velocity for the flow to the downward direction are 5m and 5m/s respectively through the circular pipe of 2m diameter. Find downstream pressure head if the headloss encountered thorough the flow path is 1.25m.



Problems on Energy Equation

A liquid of Sp Gr 0.85 is flowing through a pipe of uniform diameter. Pressure head at point A and B is 150 kN/m and 250 kN/m . Determine the flow direction and the headloss through the pipe.



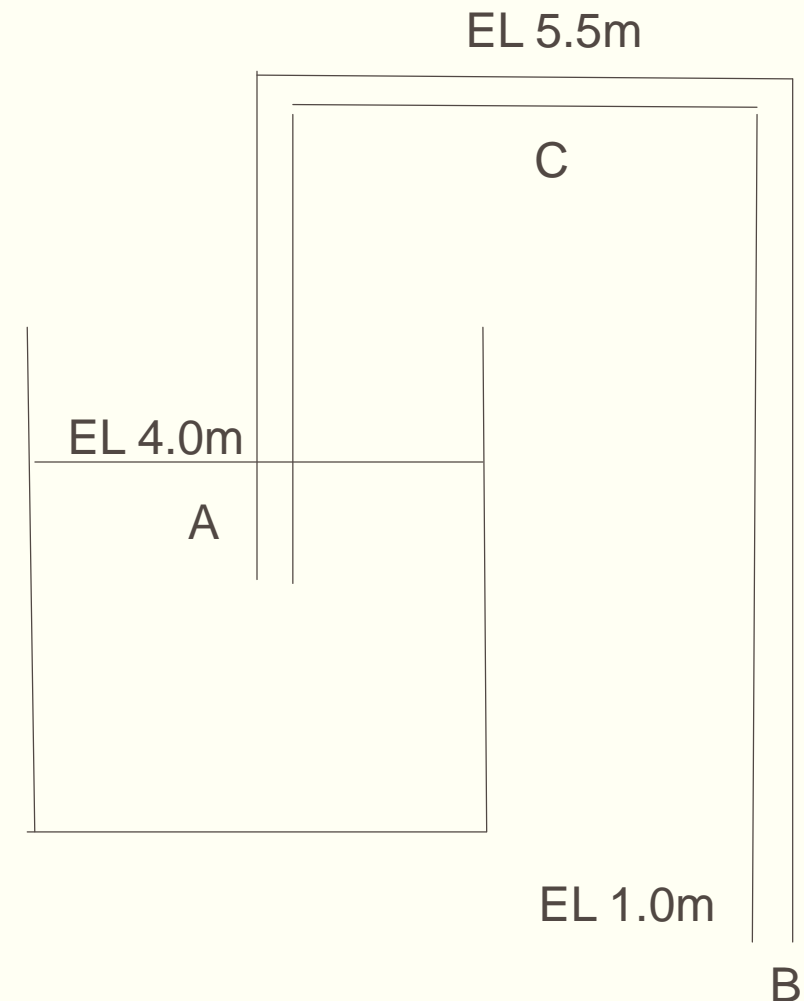
Problems on Energy Equation

Consider the flow through a syphon consisting a pipe of 15 cm diameter.

➤ HL (A-C) = 0.5m, HL (C-B) = 1.2m

➤ RD of Fluid = 0.8

Find the discharge, Q and Pressure at Point C



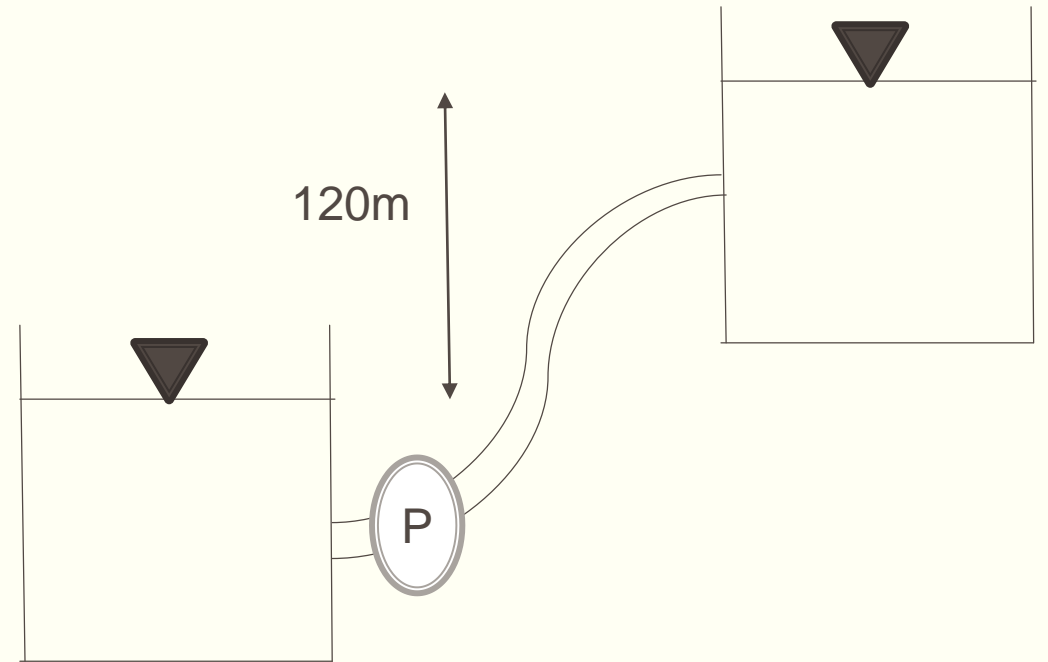
Power Consideration in Fluid Flow

$$Power = \frac{Energy}{time} = \frac{Energy}{Weight} \times \frac{Weight}{time} = H\gamma Q$$

$$1\text{HP} = 746\text{ W}$$

Problem

A pump lifts water at the rate of $6 \text{ m}^3/\text{s}$ to a height of 120m . The friction loss in the pipe is 10m . What is the horsepower required if the pump efficiency is 90%



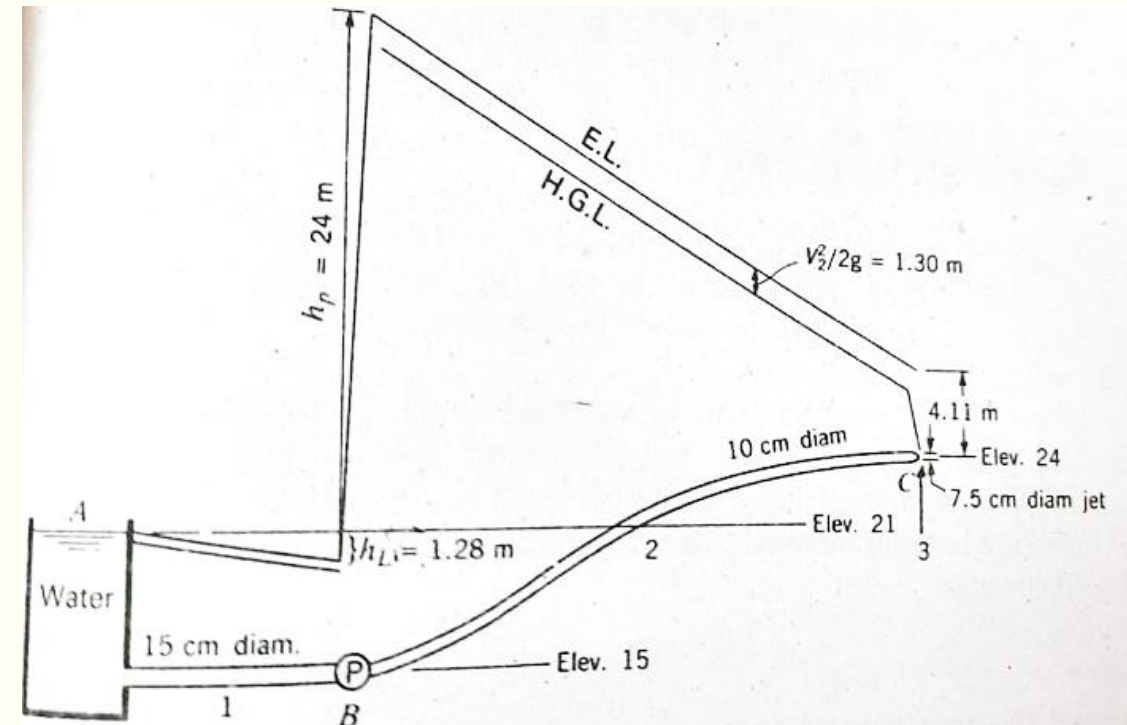
Problem

A turbine is set 40m below the water level of a reservoir and is fed by a 60cm diameter pipe. A short pipe of 45cm diameter discharges the water from the turbine to the atmosphere. Find:

- a. Neglecting friction, estimate the power extracted by the turbine when the discharge is $0.8 \text{ m}^3/\text{sec}$.
- b. If a total friction loss of 10m is assumed on the turbine efficiency is 85%, estimate the power output.

Problem

A pipeline with a pump leads to a nozzle. Find the flow rate when the pump develops a head of 24m. Assume that the head loss in the 15cm diameter pipe may be expressed by $h_L = 5 \frac{V_1^2}{2g}$, while the head loss in the 10cm diameter pipe is $h_L = 12 \frac{V_2^2}{2g}$. Sketch the energy line and hydraulic grade line, and find the pressure head at the suction side of the pump.

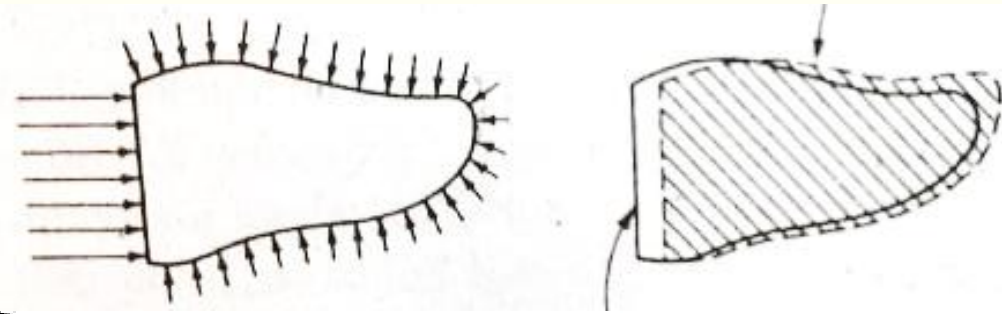




CE 261

CHAPTER 6
Momentum and Forces In Fluid Flow

Development of Impulse-Momentum principle



$(mV)_t$ = momentum at time t of the fluid system (coincident with the control volume at time t)

$(mV)_{t+\Delta t}$ = momentum at time $(t + \Delta t)$ of the fluid system (coincident with the shaded area of Fig. 6.1 at time $t + \Delta t$)

$(m'V')_t$ = momentum of the fluid mass contained within the control volume at time t

$(m'V')_{t+\Delta t}$ = momentum of the fluid mass contained within the control volume at time $(t + \Delta t)$

$\Delta(mV)_{\text{out}}$ = momentum of the fluid mass that leaves the control volume during time interval Δt

$\Delta(mV)_{\text{in}}$ = momentum of the fluid mass that enters the control volume during time interval Δt

Development of Impulse-Momentum principle

$$(mV)_{t+\Delta t} = (m'V')_{t+\Delta t} + \Delta(mV)_{\text{out}} - \Delta(mV)_{\text{in}}$$

The change of momentum of the fluid system is

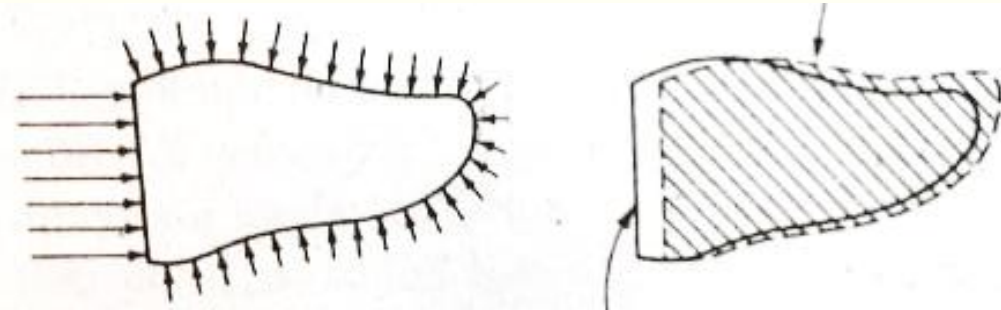
$$\Delta(mV) = (mV)_{t+\Delta t} - (mV)_t$$

Substituting the two preceding expressions into Eq. (6.2), we get

$$\Delta(mV) = (m'V')_{t+\Delta t} - (m'V')_t + \Delta(mV)_{\text{out}} - \Delta(mV)_{\text{in}}$$

Applying Eq. (6.1), dividing through by Δt , rearranging, and noting that the limit of $\Delta(mV)/\Delta t = d(mV)/dt$ as $\Delta t \rightarrow 0$, we get

$$\begin{aligned}\sum F &= \lim_{\Delta t \rightarrow 0} \frac{\Delta(mV)}{\Delta t} = \frac{d(mV)}{dt} \\ &= \frac{d(mV)_{\text{out}}}{dt} - \frac{d(mV)_{\text{in}}}{dt} + \frac{(m'V')_{t+\Delta t} - (m'V')_t}{dt}\end{aligned}\quad (6.3)$$



Development of Impulse-Momentum principle

$$\sum \mathbf{F} = \frac{d(m\mathbf{V})_{\text{out}} - d(m\mathbf{V})_{\text{in}}}{dt} = \frac{d(m\mathbf{V})_{\text{out}}}{dt} - \frac{d(m\mathbf{V})_{\text{in}}}{dt}$$

Development of Impulse-Momentum principle

$$\sum F_x = \rho_2 Q_2 V_{2x} - \rho_1 Q_1 V_{1x} = \rho Q(\Delta V_x)$$

$$\sum F_y = \rho_2 Q_2 V_{2y} - \rho_1 Q_1 V_{1y} = \rho Q(\Delta V_y)$$

$$\sum F_z = \rho_2 Q_2 V_{2z} - \rho_1 Q_1 V_{1z} = \rho Q(\Delta V_z)$$

Problems

Mathematical questions will be practiced from text books